COMFAA 3.0

An FAA Computer Program to Compute Pavement Classification Numbers (PCN)

Presented to: SWIFT 2010 Conference By: Gordon Hayhoe, FAA AJP-6312, WJHTC Date: September 13, 2010



Acknowledgements

- Rodney Joel and Jeffrey Rapol, FAA AAS-100.
- Ken DeBord and Mike Roginski, Boeing Commercial Airplane Co.
- Gary Mitchell, ACPA, and Roy McQueen.

Download AC 150/5335-5A at www.faa.gov/airports/resources/advisory_circulars

Contact Jeffrey Rapol for a copy of Draft AC 150/5335-5B at jeffrey.rapol@faa.gov

Download COMFAA 3.0 Beta at www.airporttech.tc.faa.gov/naptf/download/index1.asp



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ACN-PCN – Technical Evaluation

Basic Steps to Determine PCN in AC 150/5335-5A

- 1) Identify the features and properties of the pavement.
- 2) Determine the traffic mix.
- 3) Compute the design thickness for each aircraft alone.
- 4) The largest design thickness identifies the "critical" aircraft.
- 5) Convert traffic to equivalent traffic of the critical aircraft.
- 6) Determine the maximum allowable operating weight of the critical airplane.
- 7) Determine the ACN of the critical airplane at its maximum allowable operating weight.
- 8) Report PCN as the ACN from step 7.



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The ACN-PCN System

 Aircraft Classification Number (ACN) is precisely specified as a standard by ICAO in Annex 14 to the Convention on International Civil Aviation. Aircraft manufacturers are required to publish properly computed ACN values for all of their aircraft.



The ACN-PCN System

- Procedures for determining Pavement Classification Number (PCN) are given in the ICAO Aerodrome Design Manual, Part 3, Pavements.
- The PCN procedures in the manual are for guidance only and a great deal of latitude is provided.
- Airport operators are responsible for determining and publishing PCN values for runways.



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INTERNATIONAL STANDARDS AND RECOMMENDED PRACTICES

AERODROMES

ANNEX 14

TO THE CONVENTION ON INTERNATIONAL CIVIL AVIATION

VOLUME I AERODROME DESIGN AND OPERATIONS

THIRD EDITION - JULY 1999

This edition incorporates all amendments to Annex 14, Volume I, adopted by the Council prior to 6 March 1999 and supersedes on 4 November 1999 all previous editions of Annex 14, Volume I.

For information regarding the applicability of the Standards and Recommended Practices, *see* Chapter 1, 1.2 and Foreword.

INTERNATIONAL CIVIL AVIATION ORGANIZATION

Annex 14 to the Convention on International Civil Aviation

COMEAA 3 0 SWIFT 2010 Conference September 13, 2010



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Doc 9157-AN/901 Part 3

ICAO Aerodrome Design Manual Part 3 – Pavements

AERODROME DESIGN MANUAL



PART 3 PAVEMENTS

SECOND EDITION - 1983

Approved by the Secretary General and published under his authority

INTERNATIONAL CIVIL AVIATION ORGANIZATION

COMEDA 3 0 SWIFT 2010 Conference September 13, 2010



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Definitions in Annex 14

- ACN A number expressing the relative effect of an aircraft on a pavement for a specified standard subgrade strength.
- PCN A number expressing the bearing strength of a pavement for <u>unrestricted</u> operations.



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Therefore:

- If a particular aircraft at a given weight has an ACN less than, or equal to, the PCN of a particular pavement (ACN <= PCN), then no restrictions need to be placed on operation of that aircraft on that pavement.
- Overload evaluation is a separate topic and will not be discussed.



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PCN Reporting Information

- Pavement Type: Flexible or Rigid.
- Subgrade Strength: High, Medium, Low, or Ultra Low.
- Maximum Allowable Tire Pressure (flexible only): High, Medium, Low, Very Low.
- Pavement Evaluation Method: Using or Technical.



ACN-PCN SYSTEM – PCN Values

 PCN values are reported in a coded format using 5 parts separated by "/"

Sample 39/F/B/X/T

Information includes

- Numerical PCN Value = 39
- Pavement Type = Flexible
- Subgrade Category = 10 CBR
- Allowable Tire pressure <= 1.5 MPa = 218 psi</p>
- Method used to determine the PCN value = Using or Technical



Proposed Change in Tire Pressure Limits

Tire Pressure Category	Current ICAO designations and limits	Proposed ICAO designations and limits
W	High: no pressure limit	Unlimited
Х	Medium: pressure limited to 1.50 MPa	High : pressure limited to 1.75 MPa
Y	Low: pressure limited to 1.00 MPa	Medium: pressure limited to 1.25 MPa
Ζ	Very low: pressure limited to 0.50 MPa	Low: pressure limited to 0.50 MPa



ACN Computation

- Ratio of a computed single-wheel load to a reference single-wheel load with a tire pressure of 1.25 MPa (181 psi).
- Flexible: Based on the USACOE ESWL CBR method of design using alpha factors adopted by ICAO October 2007. Thickness is computed for 10,000 coverages.
- Rigid: PCA Westergaard interior stress method of design. Thickness is computed for 10,000 coverages at a concrete strength of 4.217 MPa (620 psi) (working stress of 2.75 MPa (400 psi)).
- <u>These are fixed standard procedures. Other</u> <u>design procedures or traffic levels cannot be</u> <u>substituted</u>.



Subgrade Strength for ACN Computation

- Flexible: The CBR of the subgrade soil.
- Rigid: The k value at the top of the support, including all subbase layers. It is not the same as the k value of the subgrade soil.



Using Aircraft Method for PCN

- Find the ACN of all of the aircraft regularly using the pavement and pick the largest ACN to be the PCN of the pavement.
- But see page 3-27 of the ICAO manual: "Support of a particularly heavy load, but only rarely, does not necessarily establish a capability to support equivalent loads on a regular repetitive basis." Where is the line between regular and overload operation?



Technical Method for PCN

- The ICAO manual covers in detail a very broad range of methods, including:
 - Any rational design procedure developed specifically for airport pavements but applied in reverse for pavement evaluation.
 - Pavement surface deflection measured under the load from a representative aircraft.
 - Non-destructive test results with backcalculation.
 - Allows for design and evaluation procedures not in use when the manual was written.



- In 3.3.3 "Since the effectiveness of aircraft undercarriages using multiple wheels is greater on pavements founded on strong subgrades compared to those on weak subgrades, the problem of reporting bearing strength is complicated."
- Hence four <u>representative</u> subgrade strengths.



- In 3.4.2 "Thus fairly large variations can exist in the loading-repetitions relation without serious differences in evaluation resulting."
- The statement is supported by the fact that the relationship between load and repetitions is logarithmic rather than direct.



- This assumption is probably the justification for ACN being computed at 10,000 coverages for all aircraft, independent of the use levels.
- This basic assumption in the ACN-PCN method appears to break down at large hub airports with a <u>very large number</u> of small aircraft departures and a relatively small number of very large aircraft departures.



- In 3.6.3.1 "From the chosen design method and established quantities for the design elements, limiting load or mass can be established for any aircraft expected to use the pavement."
- Establishes the maximum allowable load for any aircraft used in the evaluation.



- e) Reported PCN. The PCN to be reported can be determined from the aircraft loads (masses) which the evaluation has established as maximum allowable for the pavement. By using the evaluation load for one of the heaviest type aircraft using the pavement and information shown in Appendix 5, and interpolating as necessary, the PCN can be found. This can be done for a selected representative aircraft or for several aircraft for which evaluation of allowable load has been made. All such determinations should yield the same PCN value, or very nearly so. If there are large differences it would be well to recheck both the translation from the evaluation load and the evaluation. If differences are small an average or lower range value should be selected for reporting. If needed information is not
 - The second highlighted statement is not supported by case studies on some large hub airports.

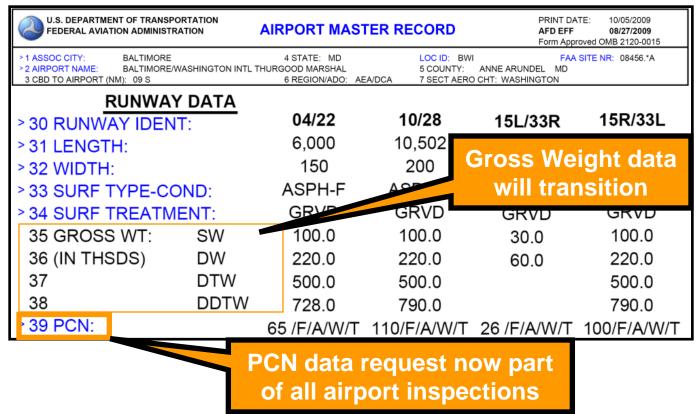


FAA Guidance on PCN Calculation

- The FAA is responsible for certifying all commercial airports in the U.S. and is the organization generally responsible for complying with international agreements on aviation.
- Well defined procedures are therefore required for determining and publishing PCN values for runways at all commercial airports in the U.S.



FAA Airport Master Record Data (Form 5010 Database)



 The Master Record is required to be updated periodically. PCN is now mandatory and Gross Weight data will possibly be phased out with time.



FAA Guidance on PCN Calculation

- 2005 The PCN field was re-activated in the Form 5010 Database.
- September, 2006 AC 150/5335-5A "Standardized Method of Reporting Airport Pavement Strength – PCN" was released to standardize the procedures for computing and reporting PCN values for inclusion in the 5010 database. (Complete re-write of AC 150/5335-5.)



FAA Guidance on PCN Calculation

- AC 150/5335-5A is based in large part on the procedures described in Boeing Report D6-82203 "Precise Methods for Estimating Pavement Classification Number," 1998.
- D6-82203 is, in turn, based in large part on recommendations contained in the ICAO Aerodrome Design Manual.



ACN-PCN – Technical Evaluation

Basic Steps to Determine PCN in AC 150/5335-5A

- 1) Identify the features and properties of the pavement.
- 2) Determine the traffic mix.
- 3) Compute the design thickness for each aircraft alone.
- 4) The largest design thickness identifies the "critical" aircraft.
- 5) Convert traffic to equivalent traffic of the critical aircraft.
- 6) Determine the maximum allowable operating weight of the critical airplane.
- 7) Determine the ACN of the critical airplane at its maximum allowable operating weight.
- 8) Report PCN as the ACN from step 7.



Draft AC 150/5335-5B

- AC 150/5335-5A was rewritten based on the work of an industry working group in which about ten case studies were evaluated (Case 1 was one).
- The new AC (-5B) is still a draft and is in the final stages of public review and comment.
- -5B is accompanied by:
 - a completely automated version of COMFAA, version 3.0.
 - and a support spreadsheet to help in deriving pavement design thickness and sorting the results.



Draft AC 150/5335-5B



Advisory Circular

Subject: STANDARDIZED METHOD OF Date: DRAFT REPORTING AIRPORT PAVEMENT STRENGTH - PCN

Initiated by: AAS-100

AC No: 150/5335-5B Change:

1. PURPOSE OF THIS ADVISORY CIRCULAR.

a. This advisory circular (AC) provides guidance for using the standardized International Civil Aviation Organization (ICAO) method to report airport pavement strength. ICAO requires member countries to report pavement strength information for a variety of purposes. The standardized method, known as the Aircraft Classification Number - Pavement Classification Number (ACN-PCN) method, has been developed and adopted as an international standard and has facilitated the exchange of pavement strength rating information.

b. The AC provides guidance for reporting changes to airport data that is generally published on Federal Aviation Administration Form 5010, Airport Master Record. The data elements associated with Gross Weight (Data Elements 35 through 38) and Pavement Classification Number (Data Element 39) are affected.

2. EFFECTIVE DATE. Effective three years after the issue date of this AC, all public-use paved runways serving aircraft with gross weights equal to or greater than 25,000 pounds at NPIAS airports must be assigned gross weight and PCN data using the guidance provided in this AC. At the issue date of this AC, about 1,850 runways met this requirement.

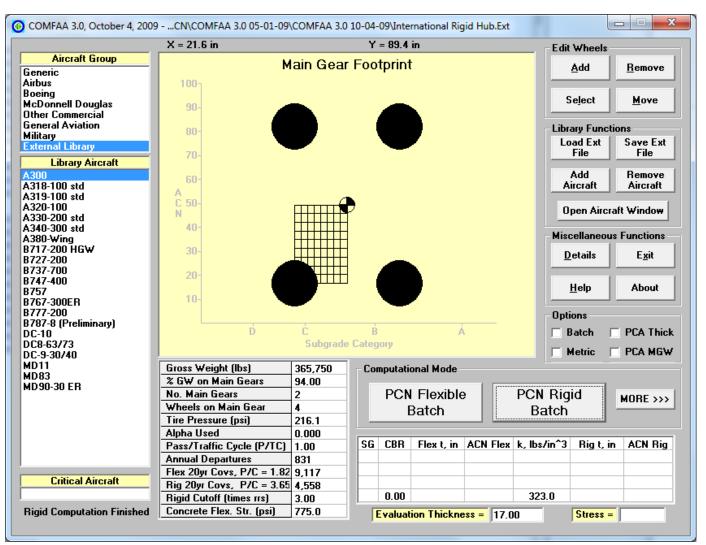


Draft AC 150/5335-5B

- The design procedures recommended in the new AC are:
 - CBR ESWL with the new alpha factors for flexible pavements.
 - Edge stress Westergaard as implemented in AC 150/5320-6C and -6D.
 - The PCA center stress method can also be selected in COMFAA 3.0.
- These were selected for backward compatibility with established methods and compatibility with the ACN computation procedure.



COMFAA 3.0 – Base Screen





The PCN Methodology

- The current methodology (-5A) finds the critical aircraft and then finds the ACN of that aircraft at the maximum allowable gross weight. That ACN is then the PCN.
- The new methodology is the same except that the ACN at maximum allowable gross weight is calculated for all of the aircraft in the mix.
- The largest ACN value is then selected as the PCN.
- There is a need for a way to eliminate "occasional or overload" aircraft from the mix.

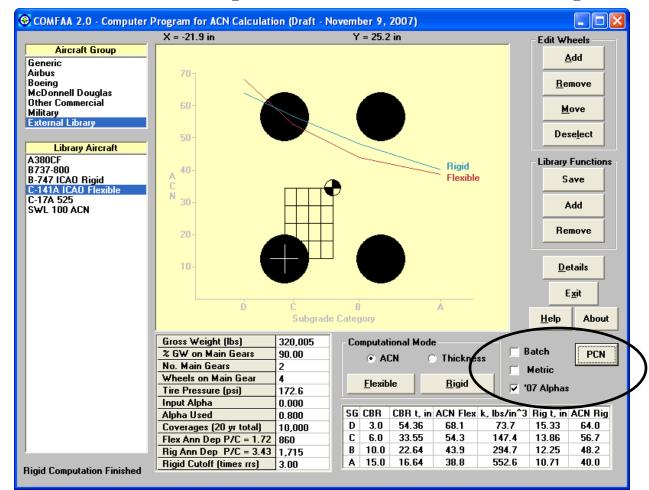


COMFAA 3.0 – Aircraft Window

Aircraft Group	Airc	raft Data3.0 05-01-09	COMFAA 3.0 10-0	4-09\Internatio	onal Rigid Hu	ub.Ext		
Generic	—	and the second	Course Franks	and the second se			_	-
Airbus								
Boeing		Aircraft	Gross	Percent	Tire	Annual	No. of Tires	Number
McDonnell Douglas	No.	Name	Weight (lbs)	GW on Gears	Press. (psi)	Departures	on Gear	of Gears
Other Commercial	1	A300	365,750	94.00	216.1	831	4	2
General Aviation	$\frac{1}{2}$	A318-100 std	124,500	90.40	147.9	654	2	2
Ailitary External Library								
	3	A319-100 std	142,500	92.60	172.6	13,002	2	2
Library Aircraft	4	A320-100	151,000	94.00	200.1	15,280	2	2
A300	5	A330-200 std	509,000	94.80	205.9	88	4	2
\318-100 std	6	A340-300 std	608,000	79.58	206.0	179	4	2
319-100 std	7	A380-Wing	1,235,000	38.05	218.0	59	4	2
A320-100 A330-200 std	8	B717-200 HGW	122,000	94.42	164.0	301	2	2
A340-300 std	9	B727-200	185,200	96.00	148.0	111	2	2
\380-Wing	10	B737-700	188,200	91.70	205.0	18,133 754 10,079	2	2
3717-200 HGW	11	B747-400	877,000	93.32	200.0		4	4
3727-200	12	B757		91.18	183.0		4	2
3737-700	13	B767-300ER	413,000	92.40	200.0	2,521	4	2
3747-400	14	B777-200	537,000	95.42	185.0	1,095	6	2
8757 8767-300ER	15	B777-200 B787-8 (Preliminary)	· ·		220.0	32	4	2
3777-200			478,000					
3787-8 (Preliminary)		DC-10	458,000	93.32 195.0 96.12 196.0 92.40 155.0	115	4	2	
)C-10		DC8-63/73	358,000		155.0	79 8	4	2
DC8-63/73	18	DC-9-30/40	109,000					
DC-9-30/40	19	MD11	633,000	77.54	206.0	44	4	2
4D11	20	MD83	161,000	94.76	195.0	739	2	2
4D83 4D90-30 ER	21	MD90-30 ER	168,500	93.96	193.0	213	2	2
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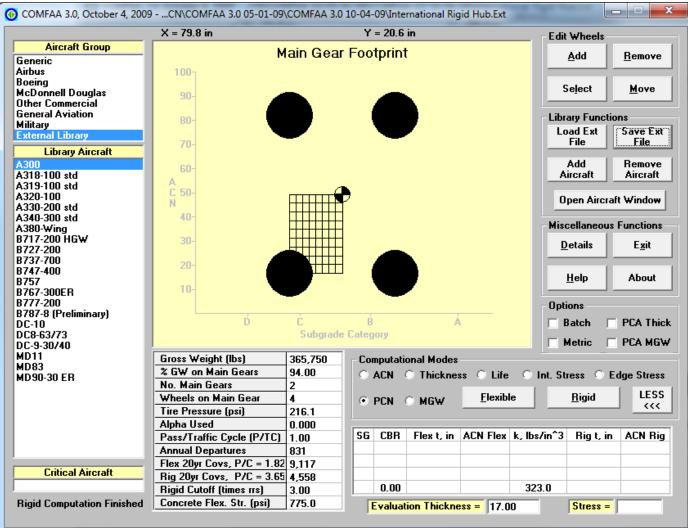


COMFAA 2.0 Will Continue to be Supported – Simpler and Old Alphas





COMFAA 3.0 – Options





COMFAA 3.0 – Sample Results

Unit Conversions Show Alpha Show Ext File Single Aircraft ACN C Other Calculation Modes C Other Calculation Modes C Desc Back 19 MD11 633,000 77.54 206.0 44 239 10.62 20 MD83 161,000 94.76 195.0 739 4,321 11.66 21 MD90-30 ER 168,500 93.96 193.0 213 1,275 11.42 Critical Thickness Maximum Aircraft Total Equiv. Covs. Equiv. Covs. Maximum Ailowable A164.4 76.1 88.0 98.4 3 A319-100 std >5,000,000 15.20 176,541 45.0 98.7 98.7 98.0 98.4 3 A319-100 std >5,000,000 15.20 176,541 45.0 98.6 100.6 120.2 4 A320-100 1,533,532 15.08 191,198 78.6 87.0 11.1 120.7 138.0 98.4 3 B319-100 std >56,000,000 15.24 1505,981 76.6 73.6 87.0 101.1) ICA	AO ACN Computation, D	etailed Output	-		-	Geogr Frankly	and a	
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6 A340-300 std 482,102 14.98 785,764 73.6 87.0 104.1 120.7 7 A380-Ming 358,352 14.95 1,558,981 76.6 91.2 108.1 123.8 8 B717-200 HGW >5,000,000 15.24 150,549 44.8 47.0 49.0 50.5 9 B727-200 321,940 14.94 237,374 66.9 71.1 74.9 77.9 10 B737-700 117,290 14.84 244,772 72.2 75.5 78.4 80.8 11 B747-400 934,194 15.03 1,076,982 70.4 84.6 99.5 112.5 12 B757 >5,000,000 15.41 314,766 41.2 49.3 57.7 64.9 13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93			496 735	14 98	657 683	73 3	86.6 103.6	120.2	
7 A380-Wing 358,352 14.95 1,558,981 76.6 91.2 108.1 123.8 8 B717-200 MGW >5,000,000 15.24 150,549 44.8 47.0 49.0 50.5 9 B727-200 321,940 14.94 237,374 66.9 71.1 74.9 77.9 10 B737-700 117,290 14.84 244,772 72.2 75.5 78.4 80.8 11 B747-400 934,194 15.03 1,076,982 70.4 84.6 99.5 112.5 12 B757 >5,000,000 15.41 314,766 41.2 49.3 57.7 64.9 13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11			482 102	14 98	785 764	73.6	87 0 104 1		
10 B737-700 117,290 14.84 244,772 72.2 75.5 78.4 80.8 11 B747-400 934,194 15.03 1,076,982 70.4 84.6 99.5 112.5 12 B757 >5,000,000 15.41 314,766 41.2 49.3 57.7 64.9 13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982		A380-Wing	358 352	14 95	1 558 981	76.6	91 2 108 1	123 8	
10 B737-700 117,290 14.84 244,772 72.2 75.5 78.4 80.8 11 B747-400 934,194 15.03 1,076,982 70.4 84.6 99.5 112.5 12 B757 >5,000,000 15.41 314,766 41.2 49.3 57.7 64.9 13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982		B717-200 HGW	>5 000 000	15 24	150 549	44 8	47 0 49 0	50 5	
10 B737-700 117,290 14.84 244,772 72.2 75.5 78.4 80.8 11 B747-400 934,194 15.03 1,076,982 70.4 84.6 99.5 112.5 12 B757 >5,000,000 15.41 314,766 41.2 49.3 57.7 64.9 13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982		B727-200	321 940	14 94	237 374	66.9	71 1 74 9	77 9	-
13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength Main Gear Pressure A(552)			117 290	14 84	244 772	72 2	75 5 78 4		
13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength Main Gear Pressure A(552)	11	B747-400	934 194	15 03	1 076 982	70 4	84 6 99 5		
13 B767-300ER 2,265,827 15.11 501,589 63.1 75.8 89.4 101.5 14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength Main Gear Pressure A(552)	12	B757	>5.000.000	15.41	314,766	41.2	49.3 57.7		
14 B777-200 >5,000,000 15.43 646,541 49.0 63.7 83.8 103.2 15 B787-8 (Preliminary) 288,854 14.93 590,004 77.4 92.2 108.3 122.6 16 DC-10 2,241,344 15.11 568,580 63.6 76.2 91.3 105.6 17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147)			2,265,827	15.11	501,589	63.1	75.8 89.4		
17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5	14	B777-200	>5.000.000	15.43	646.541	49.0	63.7 83.8	103.2	
17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5	15	B787-8 (Prelimina	rv) 288.854	14.93	590,004	77.4	92.2 108.3	122.6	
17 DC8-63/73 1,348,041 15.07 422,819 64.5 76.3 88.0 97.9 18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5	16	DC-10	2,241,344	15.11	568,580	63.6	76.2 91.3	105.6	
18 DC-9-30/40 >5,000,000 15.36 132,915 37.7 39.7 41.4 42.9 19 MD11 287,118 14.93 792,982 79.7 95.8 113.9 130.3 20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5	17	DC8-63/73	1,348,041	15.07	422,819	64.5	76.3 88.0	97.9	
20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74)	18	DC-9-30/40	>5,000,000	15.36	132,915	37.7	39.7 41.4	42.9	
20 MD83 287,599 14.93 205,211 65.9 68.5 70.8 72.6 21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74)	19	MD11	287,118	14.93	792,982	79.7	95.8 113.9	130.3	
21 MD90-30 ER 196,488 14.89 215,817 69.2 71.9 74.2 76.0 Rigid ACN at Indicated Gross Weight and Strength No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5			287,599	14.93	205,211	65.9	68.5 70.8	72.6	
No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5	21	MD90-30 ER							
No. Aircraft Name Gross % GW on Tire Weight Main Gear Pressure A(552) B(295) C(147) D(74) 1 A300 365,750 94.00 216.1 48.5 57.3 66.9 75.5		Ric	rid ACN at Indicate	d Gross Weight	and Strength				
Weight Main Gear Pressure A(552) B(295) C(147) D(74) 	No.	-		-					
		_		Gear Pressure	a A(552) B(295)) C(147)	D(74)		
	1	A300	365,750		48.5 57.3	3 66.9	75.5		
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Contact Jeffrey Rapol for a copy of Draft AC 150/5335-5B at jeffrey.rapol@faa.gov

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